CRITICAL ITEMS LIST

PROJECT: SRMS
ASS'Y HOMENCEATURE: BACK-UP

SYSTEM: <u>BACK-UP</u>
ASS'Y P/N: ______ SHEET:

FREA REF.	REV.	DRAWING REF. DESIGNATION	FATLURE MODE AND CAUSE	FATLURE EFFECT ON END ETEN	HOUR / FUNC. 2/1R RATIONALE FOR ACCEPTANCE CRITICALITY
4460	1	BACKUP JOINT SELECT SUITCH GTY-1 P/N PS-B7841-01	MODE: IMABILITY TO SELECT ANY JOINT. CAUSE(S): (1) OPEN WIPER. (2) MECHANICAL FAILURE.	LOSS OF CAPABILITY TO DRIVE JOINT IN BACKUP DRIVE. WORST CASE BACKUP INOPERATIVE. REDUNDANT PATHS REMAINING SINGLE AND DIRECT	ROTARY SWITCHES USED OM THE DAC PANEL ARE HERMETICALLY SEALED, AND OF A MATURE AND PROVEN DESIGN. THESE SWITCHES ARE IN COMMON USE ON THE ORBITER VEHICLE. THE SWITCHES ARE CONTROLLED BY ROCKWELL INTERNATIONAL SPECIFICATION NG 452-0049 AND HAVE BEEN QUALIFIED TO THE REQUIREMENTS OF THIS SPECIFICATION. ELECTRICAL CONNECTIONS TO THE SWITCH ARE ACHIEVED BY MEANS OF A MATING PAIR OF MB TYPE CIRCULAR CONNECTORS USING CRIMP STYLE CONTACTS. WIRING TO SWITCH CONNECTORS USING CRIMP STYLE CONTACTS. WIRING TO SWITCH CONNECTOR UTILIZES NICKEL PLATED COMOUCTORS WITH A POCYAMIDE INSULATION THE MIRING HARMESS IS DESIGNED TO BE CAPABLE OF SEPARATE TESTING (FOR INSULATION RESISTANCE DIELECTRIC STRENGTH, AND CONTINUITY). THIS SWITCH IS NOUNTED TO THE DAC PANEL BY MEANS OF THREE 6-32 FASTENERS. AFTER INSTALLATION AND TORQUING EACH SCREW HEAD IS STAKED TO THE PANEL USING A BLOO OF PROVY ADHESIVE. A DOCKEL PIN, INTEGRAL TO THE SWITCH BODDY, ENGAGES WITH THE PANEL TO PROVIDE ROTATION RESTRAINT. ANALYSIS OF THE BASIC PANEL TO PROVIDE ROTATION RESTRAINT. ANALYSIS OF THE BASIC PANEL STRUCTURE HAS DEMONSTRATED THAT THERE ARE NO RESONANCES IN THE RELEVANT VIBRATION TESTING OF THE DAC PANEL ASSEMBLY. APPLICATION ANALYSIS HAS EVEN VERIFIED BY VIBRATION TESTING OF THAT ADEQUATE ELECTRICAL STRESS MARGINS ARE ACHIEVED. AT THE PART LEVEL, QUALIFICATION/CERTIFICATION TESTING IS DEFINED BY ROCKWELL INTERNATIONAL SPECIFICATION MESSIBLY. SHOCK (20G-3 AKES), 25000 CYCLES ACTIVATION AT RATED DC CURRENT, LEAKAGE AT ONE ATMOSPHERE DIFFERENTIAL PRESSURE. FOR SWITCH OPERATIONAL CYCLES REFER TO TABLE 13. ALL UMITS ARE SUBJECTED TO ACCEPTANCE TESTS WHICH INCLUDE PRE-ACCEPTANCE RIM-IN, DIELECTRIC WITNSTANDING VOLTAGE, CONTACT RESISTANCE, ACCEPTANCE VIBRATION, SEAL TEST, VISUAL EXAMINATION AND FINAL PERFORMANCE TEST.
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PREPARED BY: MFMG

SUPERCEDING DATE: 28 OCT 86

APPROVED BY: ___



PROJECT: SRMS

ASS'Y NOMENCLATURE: BACK-UP

ASS'Y P/N: SHEET: 2

FMEA REF.	REV.	DRAWING REF. DESIGNATION	FATLURE MODE AND CAUSE	FATLURE EFFECT ON END ITEM	HOUR / FUNC. 2/1R CRITICALITY	RATIONALE FOR ACCEPTANCE
4460	1	BACKUP JOINT	MODE:	LOSS OF CAPABILITY TO	ACCEPTANCE TES	is
	•	SWITCH GTY-1 P/N PS-87841-01	SELECT ANY JOINT.	DRIVE JOINT IN BACKUP DRIVE.	THE HARDWARE I ENVIRONMENTAL	TEM IS SUBJECTED TO THE FOLLOWING ACCEPTANCE TESTS AS PART OF THE D&C PANEL ASSEMBLY.
1		12 2.00.	CAUSE(S): (1) OPEN	WORST CASE	O VIBRATION:	LEVEL AND DURATION - REFERENCE TABLE 1
			ULPER. (2) NECHANICAL	BACKUP INOPERATIVE.	O THERMAL:	+110 DEGREES F TO PLUS 10 DEGREES F (2 CYCLES - 9.5 HRS/CYCLE.)
			FAILURE.	REDUNDANT PATHS REMAINING	SYSTEM TESTS (ASSEMBLY IS FURTHER TESTED AS PART OF THE RMS TP518 RMS STRONGBACK TEST AND TP552 FLAT FLOOR RTF1ES THE ABSENCE OF THE FAILURE MODE.
				SINGLE AND DIRECT	QUALIFICATION	TESTS
					PANEL ASSEMBLY	M HAS BEEN QUALIFIED FOR ORBITER USE. THE D&C THAS BEEN SUBJECTED TO THE FOLLOWING TEST ENVIRONMENTS.
			!		O VIBRATION:	LEVEL AND DURATION - REFERENCE TABLE 1
					O SHOCK:	20G/11 MS - 3 ANES (6 DIRECTIONS)
					O THERMAL:	130 DEGREES F TO -23 DEGREES F (12 HRS PER CYCLE) (6 CYCLES)
					O HUMIDITY:	95% (120 DEGREES F TO B2 DEGREES F CYCLE IN 16 HRS) 10 CYCLES TOTAL.
					O EMC:	MIL-STD-461 AS MODIFIED BY SL-E-0002 (TEST CED1, CE02, CE03, CS01 (DC/AC), CE03, CS01 (DC/AC), CS02, CS06, RE02 (B/N), RS02, RS03, RS04)
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					PDRS OPS CHECK	: LIST (ALL VEHICLES) JSC 16987
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REPARED BY: NEWG	SUPERCEDING DATE: 28 OCT 86	APPROVED BY:	DATE:
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CRITICAL ITEMS LIST

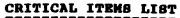
PROJECT: SRMS

ASS'Y MOMERICATURE: BACK-UP

ASS'Y P/N:

REF. REV.	DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FATLURE EFFECT ON END ITEM	HOUR / FUNC. 2/1R RATIONALE FOR ACCEPTANCE CRITICALITY
4460 1	BACKUP JOINT SELECT SWITCH GTY-1 P/N PS-87841-01	MODE: INABILITY TO SELECT ANY JOINT. CAUSE(S): (1) OPEN WIPER. (2) NECHANICAL FAILURE.	LOSS OF CAPABILITY TO ORIVE JOINT IN BACKUP DRIVE. WORST CASE BACKUP INOPERATIVE. REDUNDANT PATHS REMAINING SINGLE AMD DIRECT	REMETICALLY SEALED ROTARY SUITCHES ARE PROCURED TO ROCKWELL SPECIFICATION HC452-0049, AS REQUIRED BY CAE SPEC. PS87841. CAE PART MO. PS87841-1. QUALIFICATION AND ACCEPTANCE TESTING OF SWITCHES IS PERFORMED TO RI. SPEC. MC452-0049. RECEIVING INSPECTION VERIFIES THAT SWITCHES RECEIVED ARE AS IDENTIFIED IN THE PROCUREMENT DOCUMENTS, THAT MO PHYSICAL DANAGE HAS OCCURRED TO SWITCHES DURING SHIPMENT, THAT THE RECEIVING DOCUMENTS PROVIDE ADDICANTS THAT MO PHYSICAL DANAGE HAS OCCURRED TO SWITCHES DURING SHIPMENT, THAT THE RECEIVING DOCUMENTS PROVIDE ADDICANTS THAT MORMATION AND ACCEPTANCE TEST DATA IDENTIFIES ACCEPTABLE PARTS. PARTS ARE INSPECTED THROUGHOUT MANUFACTURE AND ASSEMBLY AS APPROPRIATE TO THE MANUFACTURING STAGE COMPLETED. THESE INSPECTIONS INCLUDE, COMPONENT MOUNTING TO FRONT PANEL INSPECTION, SOLDERING OF WIRES TO SWITCH CONTACTS, WIRE ROLITING STRESS RELIEF OF WIRES ETC., OPERATORS AND INSPECTIONS ARE TRAILNED AND CERTIFIED TO MASA WHO STORMS AND ASSEMBLY AS MODIFIED BY JSCOBBOODA. PRE-TEST INSPECTION OF DRC PANEL ASSY INCLUDES AN AUDIT OF LOWER TIER INSPECTION COMPLETION, AS BUILD CONFIGURATION VARIDATION TO AS DESIGN ETC. (SPAR/GOVERNMENT REP. MANDATORY INSPECTION POINT) A TEST READINESS REVIEW (IRR) WHICH INCLUDES VERIFICATION OF TEST PERSONNEL TEST DOCUMENTS, TEST EQUIPMENT CALIBRATION/VALIDATION STATUS AND HANDWARE CONFIGURATION IS CONVENED BY QUALITY ASSURANCE IN COMPLETING, WITCH ENGINEERING, WAS PRICTARED OF ANY FORMAL TESTING (ACCEPTANCE OR QUALIFICATION). ACCEPTANCE TESTING (ATP) INCLUDES AMBIENT PERFORMANCE, THE CONTACTS ETC. SUB-SYSTEMS INTEGRATION, THE INTEGRATION OF MECHANICAL ARM SUBASSEMBLIES AND THE FLIGHT CABLE VERTEFICATION, CONNECTOR INSPECTION FOR BENT OR PUSHBACK CONTACTS ETC. SUB-SYSTEMS INTEGRATION, THE INTEGRATION OF MECHANICAL ARM SUBASSEMBLIES AND THE FLIGHT CABLE VERTEFICATION WHICH INCLUDES GROUNDING CHECKS, INTER CONT

PREPARED BY: HFHG SUPERCEDING DATE: 28 OCT 86 APPROVED BY: RMS/BACK-UP 28



PROJECT: SRNS ASS'Y NOMENCLATURE: BACK-UP

SYSTEM: BACK-UP ASS'Y P/N:

SHEET: ___4 DRAWING REF. FAILURE MODE FAILURE EFFECT HOUR / FUNC. REV. REF. AND 2/1R RATIONALE FOR ACCEPTANCE DESIGNATION CAUSE END LIEN CRITICALITY 4460 1 BACKUP JOINT HODE: LOSS OF FAILURE HISTORY SELECT INABILITY TO CAPABILITY TO SWITCH GTY-1 SELECT ANY DRIVE JOINT IN P/N .THIOL BACKUP DRIVE. NO EEE PARTS FAILURES HAVE OCCURRED SUBSEQUENT TO ASSEMBLY OF PS-87841-01 PARTS. CAUSE(S): WORST CASE (1) OPEN WIPER. THE FOLLOWING FAILURE ANALYSIS REPORT(S) ARE RELEVANT: BACKUP MECHANICAL INOPERATIVE. FAILURE. REDUNDANT PATHS FAR 1010: REMAINING S/N 201 NOV 81 SINGLE AND DESCRIPTION DIRECT SJ YAW FAILED TO OPERATE IN BACK-UP HODE. FOUND DEFECTIVE D&C PANEL CABLE. CORRECTIVE ACTION REPAIR CABLE

PREPARED BY: MFWG

SUPERCEDING DATE: 28 OCT 86

APPROVED BY: ___

DATE:

CRITICAL ITEMS LIST

PROJECT: SRMS ASS'Y NOMENCLATURE: BACK-UP SYSTEN: BACK-UP
ASS'Y P/N:

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A4400 1 MARCHIP JOINT SELECT ANY JOINT SELECT ANY JOINT SELECT ANY JOINT MARILITY TO CARDIC ARM JOINT MARILITY TO CARDIC THE ARM ARM CAN BE JETTISONED. MORE AND DIRECT MARILIME. BERNINGHAM PATHS REDUNDANT PATHS REMAIN ARM CAN BE JETTISONED. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CAMNOT BE SAFETY DEFINING. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MADE AND THE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MADE AND THE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MADE AND THE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MADE AND THE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT ARM SMOULD NOT BE MADE AND THE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT MOME TO THE MADE AND THE ARM ON JETTISON CAMNOT BE ARM ON JETTISON. CREW TRAINING MOME MISSION CONSTRAINT MOME TO THE ARM ON JETTISON	FREA REF. R	EV.	NAME, QTY, E DRAWING RÉF. DESIGNATION	FATLURE MODE AND CAUSE	FATLURE EFFECY ON END ITEM	HOUR / FUNC. 2/1R RATIONALE FOR ACCEPTANCE CRITICALITY
1 1 1			SELECT SWITCH OTY-1 P/N	IMABILITY TO SELECT ANY JOINT. CAUSE(S): (1) OPEN WIPER. (2) MECHANICAL	CAPABILITY TO DRIVE JOINT IN BACKUP DRIVE. WORST CASE BACKUP INOPERATIVE. REDUNDANT PATHS REMAINING SINGLE AND DIRECT	LOSS OF MEXT REDUNDANT PATH RESULTS IN BEING ONE FAILURE AWAY FROM IMABILITY TO CRADIE ARM. JOINT WILL NOT DRIVE IN BACKUP ONCE PRIMARY MODES HAVE FAILED, THE BACKUP STANDBY SYSTEM WILL MOT PROVIDE THE CAPABILITY TO CRADIE THE ARM. ARM CAN BE JETTISOMED. CREW ACTION PERFORM AN EVA TO STOW THE ARM OR JETTISON. CREW TRAINING NONE MISSION CONSTRAINT ARM SHOULD NOT BE MANEUVERED TO POSITION WHERE JETTISON CANNOT BE SAFELY PERFORMED. SCREEN FAILURES 8: N/A (STANDBY REDUNDANT) OMRSD OFFLINE OPERATE BACKUP DIRECT DRIVE SWITCH TO + POSTION. VERIFY ENABLE COMMAND VOLTAGE FOR SELECTED JOINT AT DEC PANEL INTERFACE. OMRSD OFFLINE INSTALLATION OPERATE BACKUP DIRECT DRIVE SWITCH TO + POSTION. VERIFY ENABLE COMMAND VOLTAGE FOR SELECTED JOINT AT LONGERON INTERFACE. OMRSD OFFLINE INSTALLATION OPERATE BACKUP DIRECT DRIVE SWITCH TO + POSTION. VERIFY ENABLE COMMAND VOLTAGE FOR SELECTED JOINT AT LONGERON INTERFACE.

PREPARED BY: MFMG

SUPERCEDING DATE: 28 OCT 86

APPROVED BY: _____

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